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⁵Meirovitch, L., "A New Method of Solution of the Eigenvalue Problem for Gyroscopic Systems," *AIAA Journal*, Vol. 12, Oct. 1974, pp. 1337-1342.

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⁷Meirovitch, L., "On the Reduction of the Eigenvalue Problem for Spinning Axisymmetric Structures," AIAA Paper 75-159, Pasadena, Calif., Jan. 1975, to be published in *AIAA Journal*.

Errata

Large Space Telescope Oscillations Induced by CMG Friction

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IN the article, the captions for Figs. 2 and 4 are reversed.

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Index categories: Spacecraft Attitude Dynamics and Control; Navigation, Control, and Guidance Theory.

Low Reynolds Number Effect on Hypersonic Lifting Body Turbulent Boundary Layers

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IN the Nomenclature, the following changes should be made:

 $\ell = mixing length$

 y_r = characteristic thickness of boundary layer

Equations (1-3) and (5) should read:

$$\ell = ky D, \text{ for } 0 < y \le \lambda y_{\ell} / k \tag{1}$$

$$\ell = \lambda y_{\ell}, \text{ for } \lambda y_{\ell} / k < y \tag{2}$$

$$\ell = \mathcal{L} \lambda y_{\ell}, \text{ for } \mathcal{L} \lambda y_{\ell} / k < y \tag{3}$$

$$\delta_w^+ = (\rho_w V_{\tau, w} y_\ell) / \mu_w \tag{5}$$

The symbol y_{ε} for the characteristic thickness of the boundary layer should be changed to y_{ε} in the first two paragraphs of the Analysis section.

Reference 6 should read:

⁶McDonald, H., "Mixing Length and Kinematic Eddy Viscosity in a Low Reynolds Number Boundary Layer," Rept. J214453-1, Sept. 1970, Research Lab., United Aircraft Corp., East Hartford, Conn.

Index categories: Boundary Layers and Convective Heat Transfer—Turbulent; Supersonic and Hypersonic Flow.

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